

Micropump for MON-25/MMH Propulsion and Attitude Control, Phase I

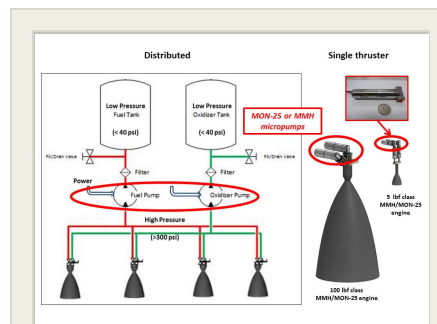
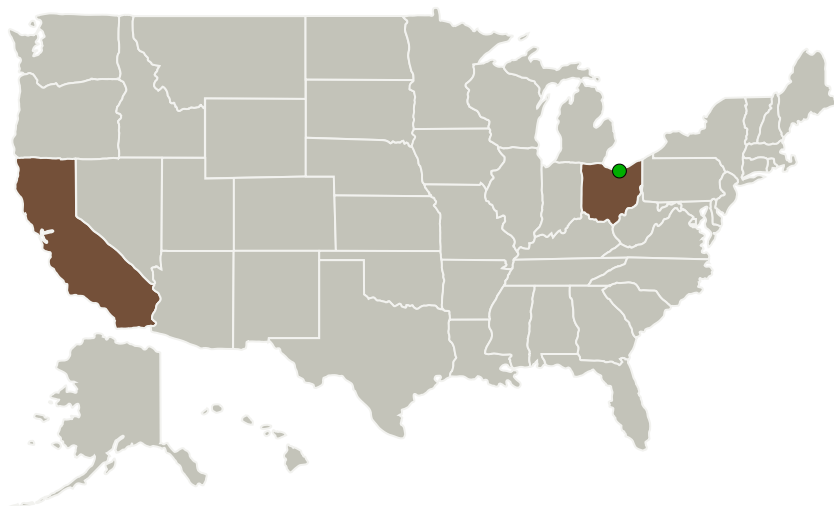
Completed Technology Project (2015 - 2015)



Project Introduction

Flight Works is proposing to expand its work in micro-gear-pumps for hypergolic and "green" propellants and team with Aerojet-Rocketdyne in order to develop and demonstrate a micropump for MON-25 and mono methyl hydrazine (MMH) bipropellant thrusters. MON-25, with 25% of nitric oxide (NO) and 75% nitrogen tetroxide (NTO, N₂O₄), allows lowering the oxidizer freezing point to -55 C, which is a close match to that of the fuel, MMH (which is around -51 C). While toxic, this propellant combination is hypergolic and allows operations over a wide range of temperatures, particularly in extremely cold environments as those envisioned for many future missions. The introduction of a micropump in the propulsion system provides many benefits, including the elimination of the pressurization systems; lighter, cheaper, and conformal tanks; improved system packaging; removal of propellant cross-contamination in the pressurization system; and long term storage for extended duration missions (since the loss of helium is no longer a concern). Under a Phase I SBIR, Flight Works Inc. is prepared to develop and characterize a micropump suitable for both MMH and MON-25, initially sized for 22-30 N (5-7 lbf) class thrusters with approximately 2.5 MPa (365 psi) inlet pressure, with the goal of demonstrating the technology with pump-fed MMH/MON-25 hot fire tests by the end of Phase II.

Primary U.S. Work Locations and Key Partners



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Organizations Performing Work	Role	Type	Location
Flight Works, Inc.	Lead Organization	Industry	Irvine, California
● Glenn Research Center(GRC)	Supporting Organization	NASA Center	Cleveland, Ohio

Primary U.S. Work Locations

California	Ohio
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Project Transitions

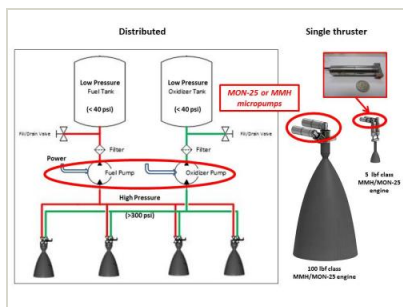
**June 2015:** Project Start**December 2015:** Closed out

Closeout Summary: Micropump for MON-25/MMH Propulsion and Attitude Control, Phase I Project Image

Closeout Documentation:

- Final Summary Chart Image(<https://techport.nasa.gov/file/139101>)

Images



Briefing Chart Image

Micropump for MON-25/MMH Propulsion and Attitude Control, Phase I
(<https://techport.nasa.gov/image/126187>)

Organizational Responsibility

Responsible Mission Directorate:

Space Technology Mission Directorate (STMD)

Lead Organization:

Flight Works, Inc.

Responsible Program:

Small Business Innovation Research/Small Business Tech Transfer

Project Management

Program Director:

Jason L Kessler

Program Manager:

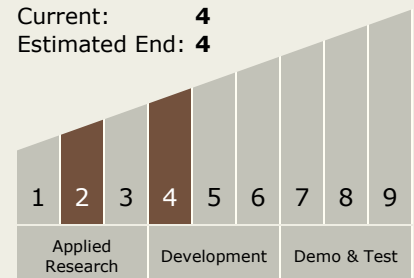
Carlos Torrez

Principal Investigator:

Nadim R Eid

Technology Maturity (TRL)

Start: 2
Current: 4
Estimated End: 4



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Technology Areas

Primary:

- TX01 Propulsion Systems
 - └ TX01.1 Chemical Space Propulsion
 - └ TX01.1.2 Earth Storable

Target Destinations

The Sun, Earth, The Moon, Mars, Others Inside the Solar System, Outside the Solar System